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# Use of flexible polymers (adhesives) in the design of aerospace applications

F KADIOGLU

**University of Turkish Aeronautical Association**

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# Frame of the presentation

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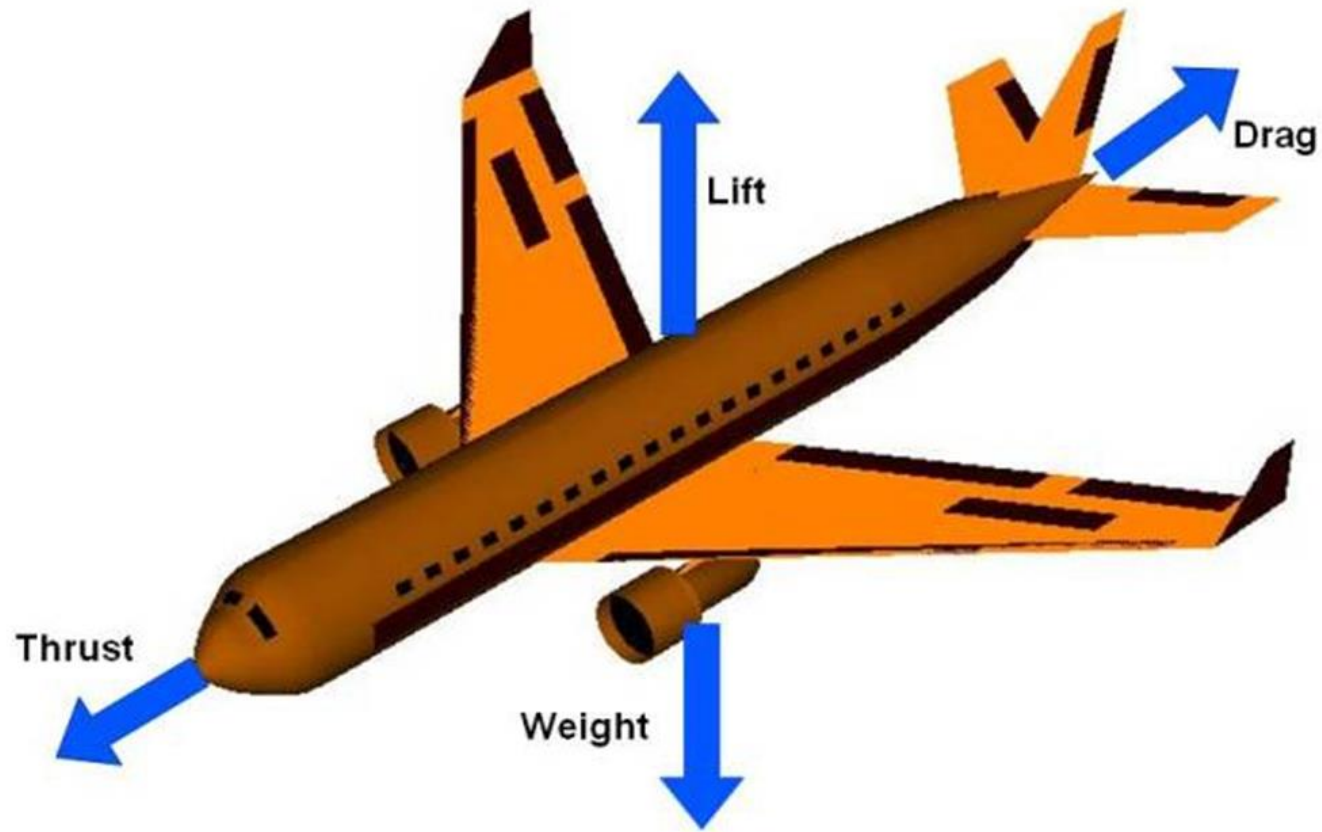
- Why polymer matrix composite materials?
- Some basic information about adhesives & adhesive joints
- Two examples

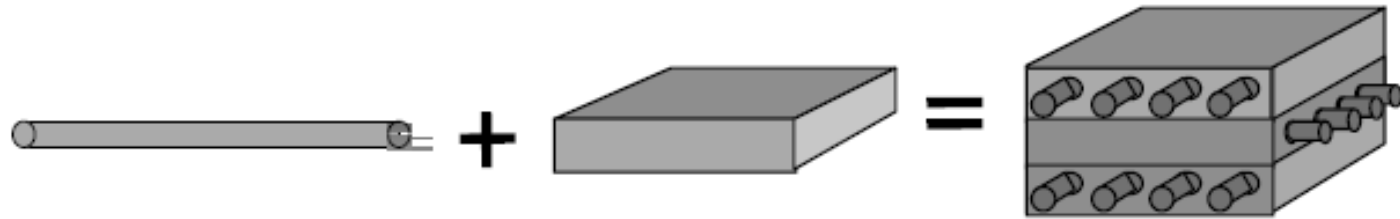
# Why polymer matrix composite materials?

National Aeronautics and Space Administration



## *Four Forces on an Airplane*





### **Fiber/Filament Reinforcement**

- High strength
- High stiffness
- Low density

### **Matrix**

- Good shear properties
- Low density

### **Composite**

- High strength
- High stiffness
- Good shear properties
- Low density

# Polymer Matrix Composite Materials

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- Composites can be very **strong** and **stiff**, yet very **light** in weight, so ratios of **strength-to-weight** and **stiffness-to-weight** are several times greater than steel or aluminum
- **Fatigue** properties are generally better than for common engineering metals
- **Toughness** is often greater too
- Composites can be designed that do not **corrode** like steel
- Possible to achieve combinations of properties not attainable with metals, ceramics, or polymers alone

# Disadvantages and Limitations of Composite Materials

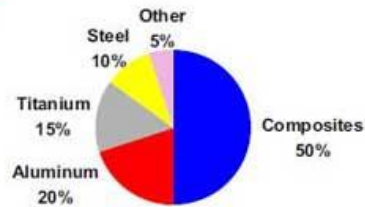
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- Properties of many important composites are **anisotropic** - the properties differ depending on the direction in which they are measured – this may be an advantage or a disadvantage
- Many of the polymer-based composites are subject to attack by **chemicals or solvents, temperature, humidity**, just as the polymers themselves are susceptible to attack
- Composite materials are generally **expensive**
- Manufacturing methods for shaping composite materials are often **slow and costly**

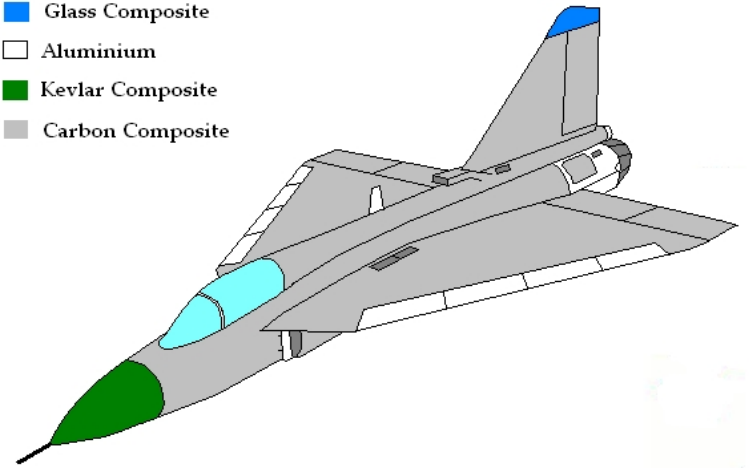
# Applications



- Carbon laminate
- Carbon sandwich
- Fiberglass
- Aluminum
- Aluminum/steel/titanium pylons



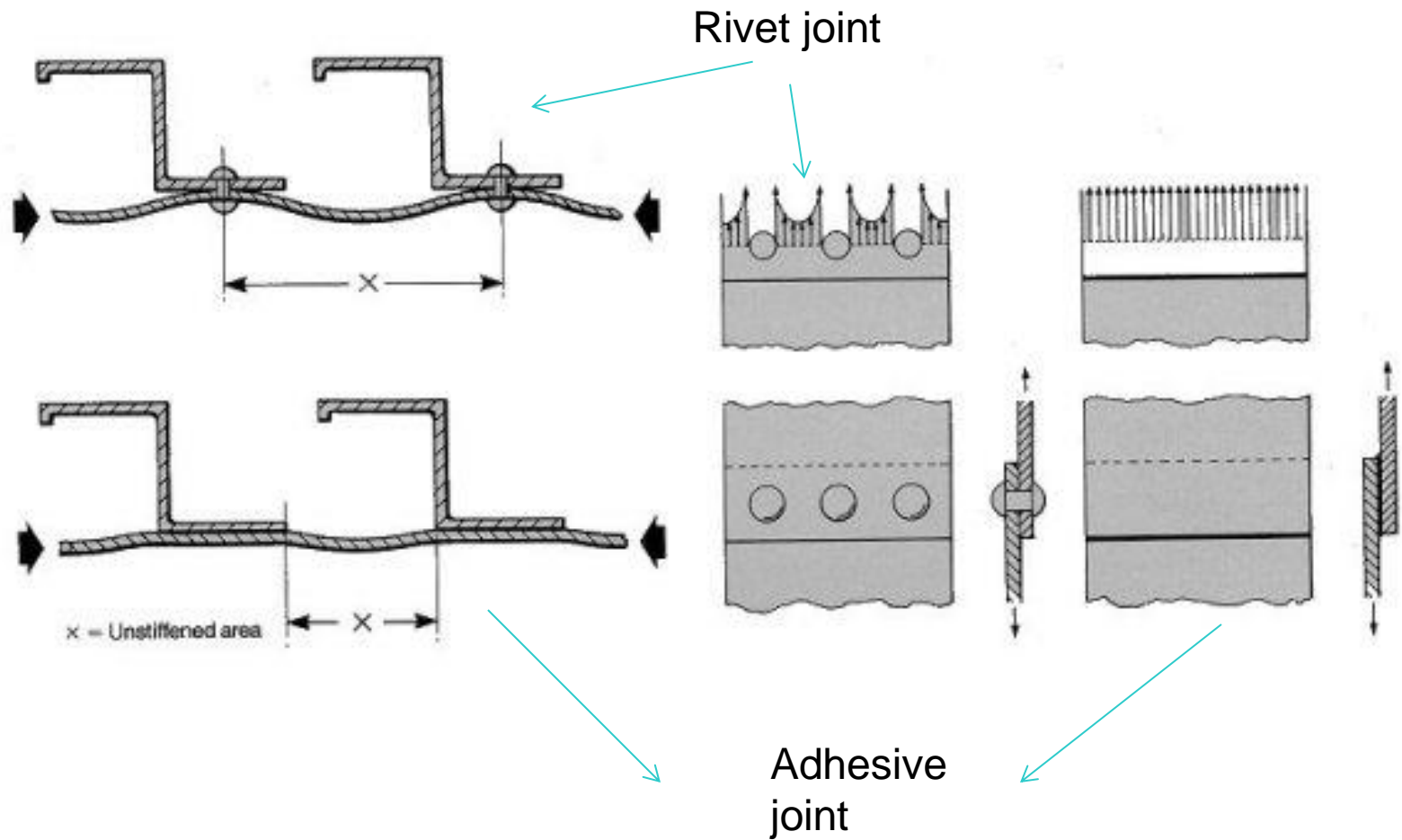
- Glass Composite
- Aluminium
- Kevlar Composite
- Carbon Composite



Composites in LCA



# Joining techniques of composites





# Types of adhesive joints

(a) Single Lap



(b) Double Lap



(c) Butt Strap



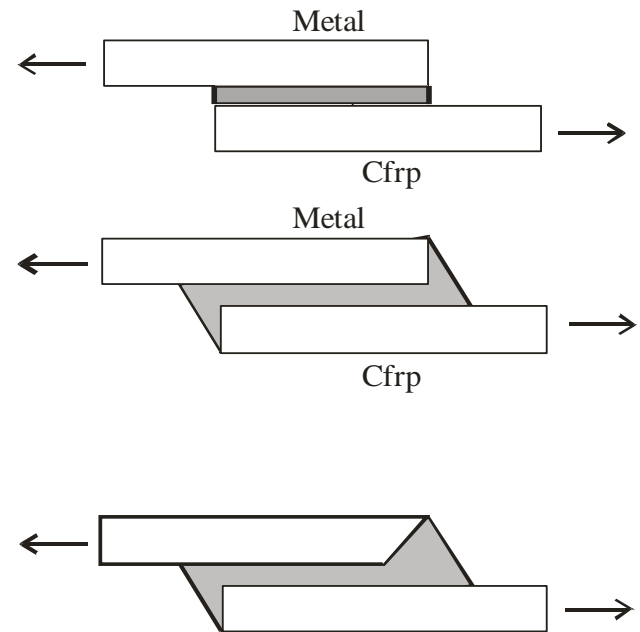
(d) Step



(e) Bevel



(f) Scarf



# Definition of an adhesive

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- An **adhesive** is defined as a substance capable of holding materials together by surface attachment.
  - **Structural Adhesives**: usually reckoned to be those with a high strength (50 MPa and upwards) and (these days), a strain to failure of at least 10% in tension, and which usually have a tensile modulus of 2 GPa or so
  - **Flexible Adhesives**: significantly less stiff, less strong, but much more ductile

# Advantages

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- Joining **dissimilar materials** (dissimilar thin sheets and foils)
- Improvement in the **appearance of the finished assembly** by the elimination of irregular surface
- the fabrication of complex shapes
- Relatively **uniform distributions of stress** over the entire bonded area
- Improving **fatigue resistance** and giving good **vibration damping**
- .....

# Disadvantages

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- **residual stresses** in bonded joints (different thermal expansion)
- curing requirement
- sensitivity to **peel loading, creep** etc...
- degradation of the adhesive caused by **heat, cold, chemical agents**, radiation, and bio-deterioration
- not easily dismantled for repair or salvage
- assessment of bond quality

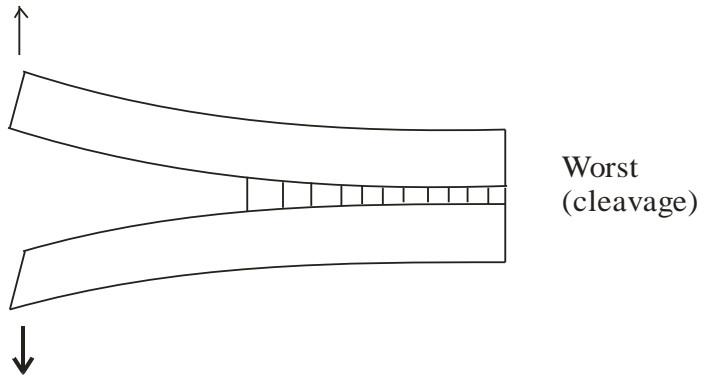
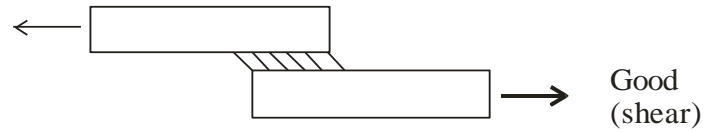
# The most important parameter! Surface pretreatment

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- **Chemical solutions** to clean the surfaces
- **Surface modification methods**

# Loading types

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# Failure mechanisms

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- Adhesive failure
  - Poor surface treatment (failure in adhesive-adherend interlayer)
- Cohesive failure
  - able to get mechanical performance of an adhesive (failure in adhesive or in adherend)
- Adhesive-Cohesive failure
  - usually due to the adherend yielding

# Design- Predictions

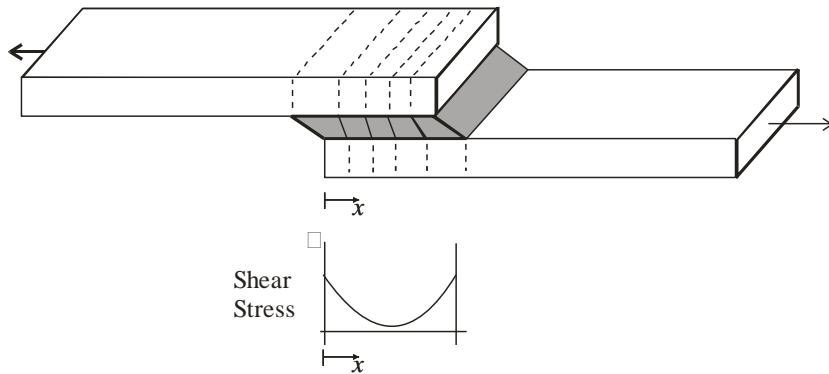
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- Analytical Analysis
  - Volkersen's elastic adherend model
  - Goland & Reissner bending model
- Numerical Analysis
  - FEM (the most effective tool)

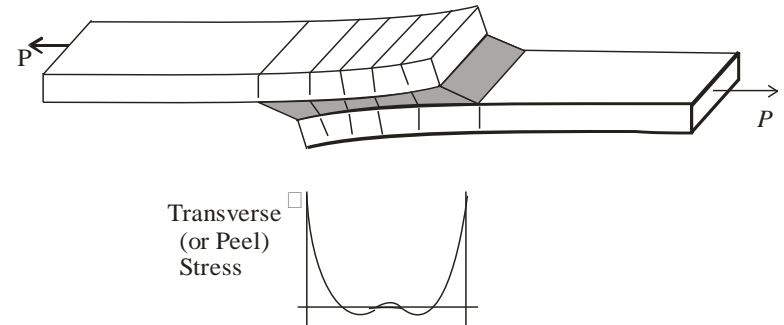


# Problems with analytical analysis

- Volkersen's elastic adherend model



- Goland & Reissner bending model



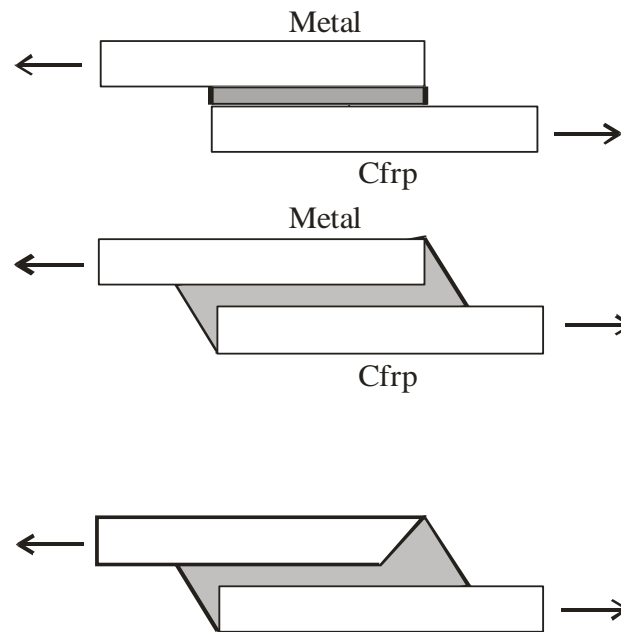
# Reality and three-dimensional stress analysis

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- differential straining of the adherends causing a shear stress distribution in the adhesive which is a maximum at the ends of the overlap (Volkersen (1938))
- offset loading of the lap joint which causes the loaded adherend to bend adjacent to the overlap region (Goland and Reissner (1944))
- end effects such as the adhesive free surface, spew, and material and geometric discontinuities which all affect the stresses at the overlap ends (Crombe and Adams (1981))
- Poisson's ratio effects
- adherend plasticity

# Some precautions to reduce the stress concentrations at the ends

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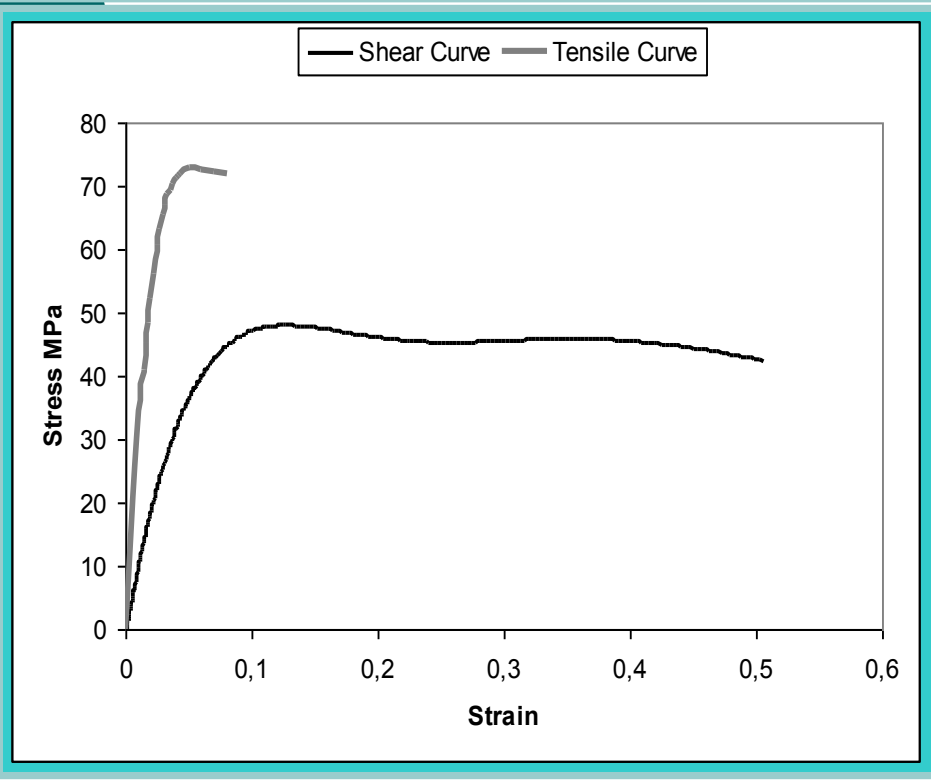


## Examples

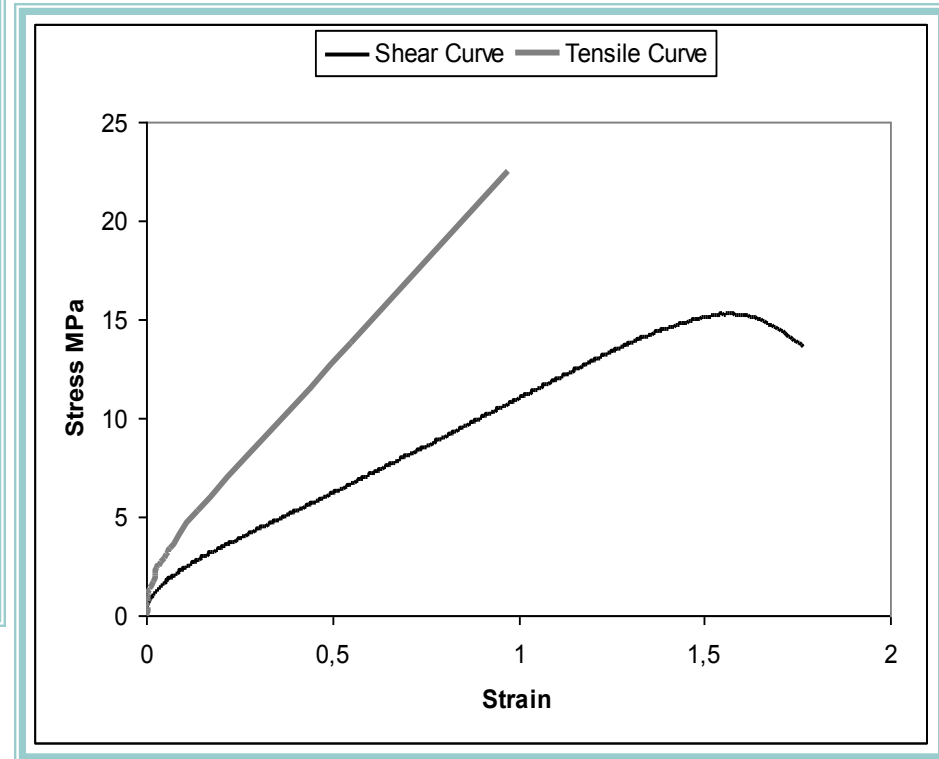
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- **A Structural Adhesive: AV119**, one-part epoxy, manufactured by Ciba Polymers. cured at 120 C for an hour according to manufacturer's instructions.
- **A Flexible Adhesive: 9245 Structural Bonding Tape (SBT)**, mixture of acrylic and epoxy, cured at 140 C for 45 minutes according to the manufacturer's instructions.

# Mechanical behavior



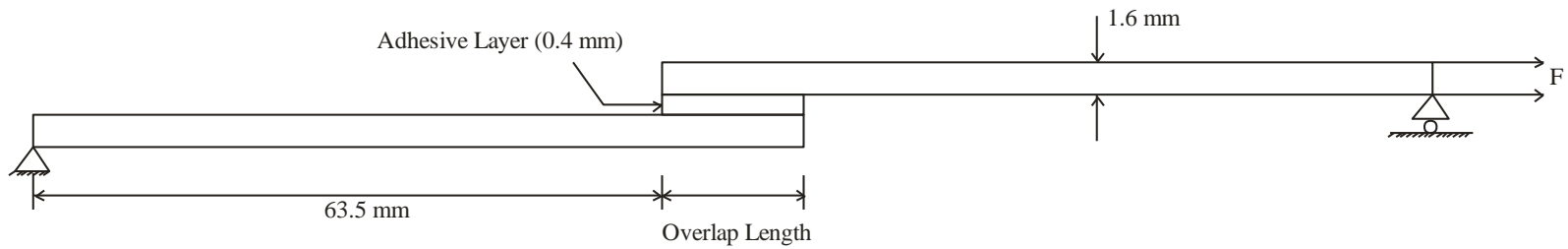
Structural Adhesive AV119



Flexible Adhesive SBT

# Lap shear performance in quas-static loading under tensile loading

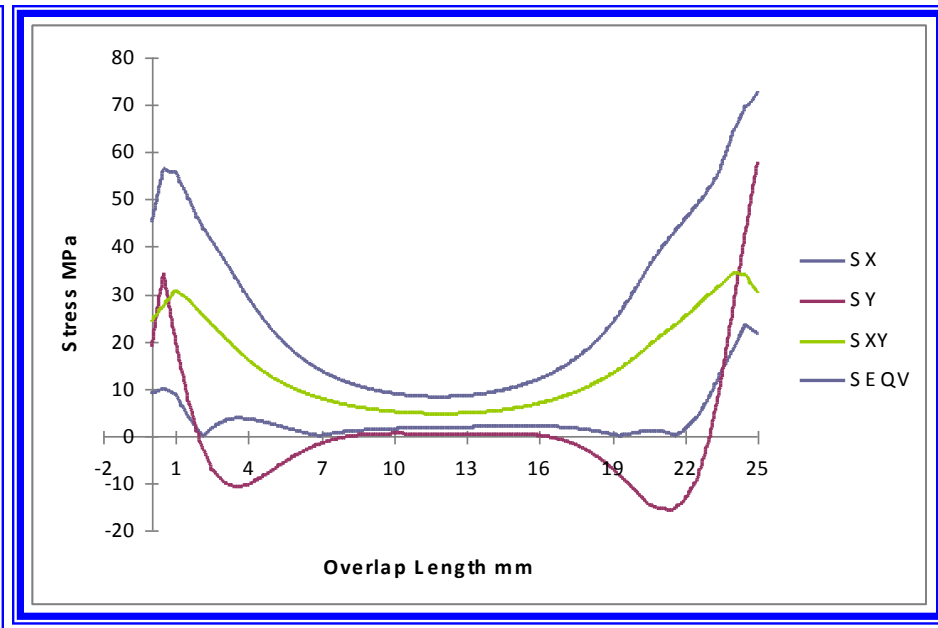
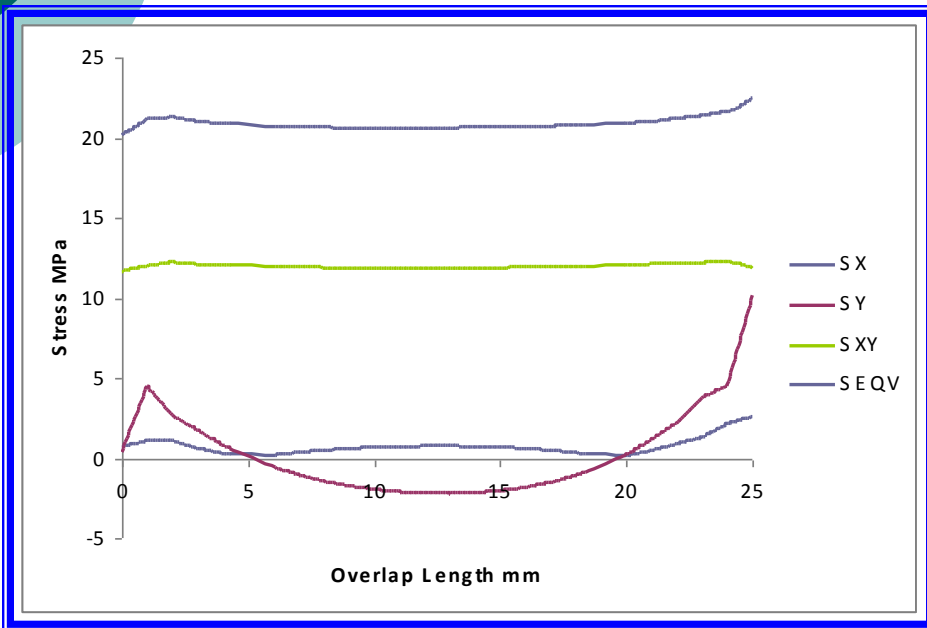
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# Stress distributions in the adhesive layers under tensile loading

- SBT (Flexible Adhesive)

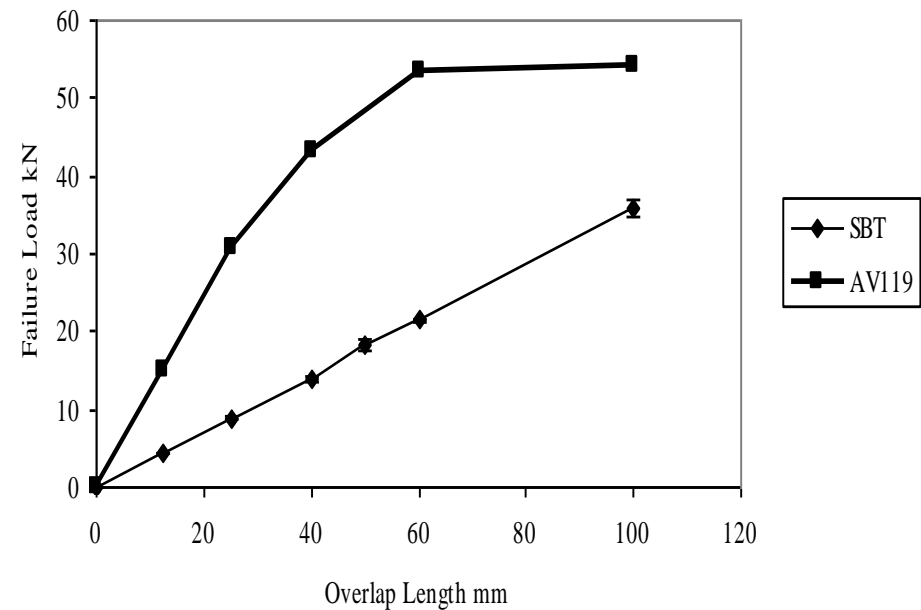
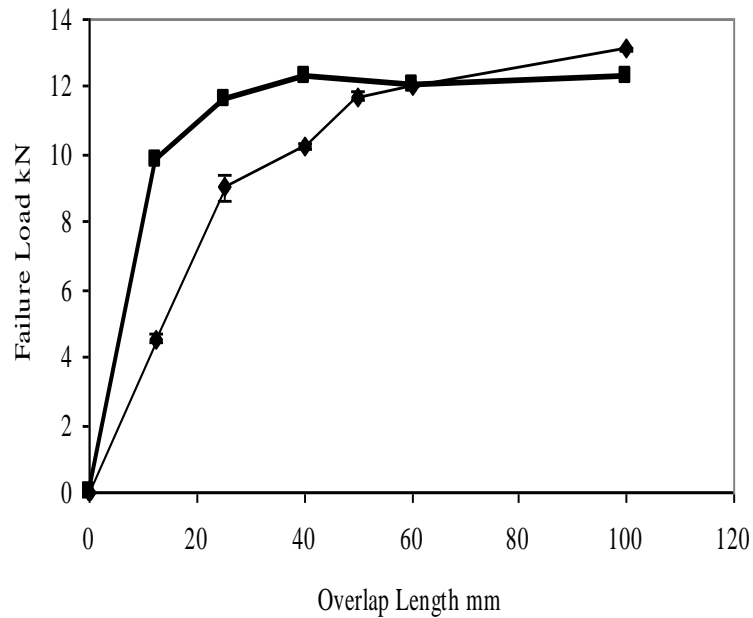
- AV119 (Structural Adhesive)



# Comparison of results from tensile loading

○ Mild steel adherends

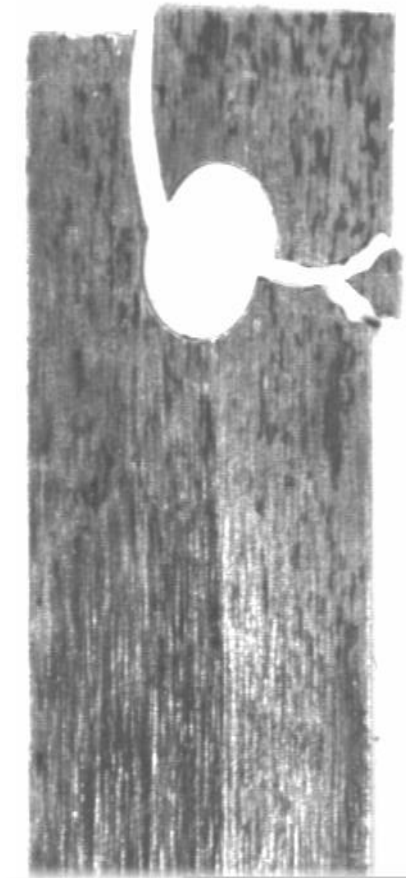
○ Hard steel adherends



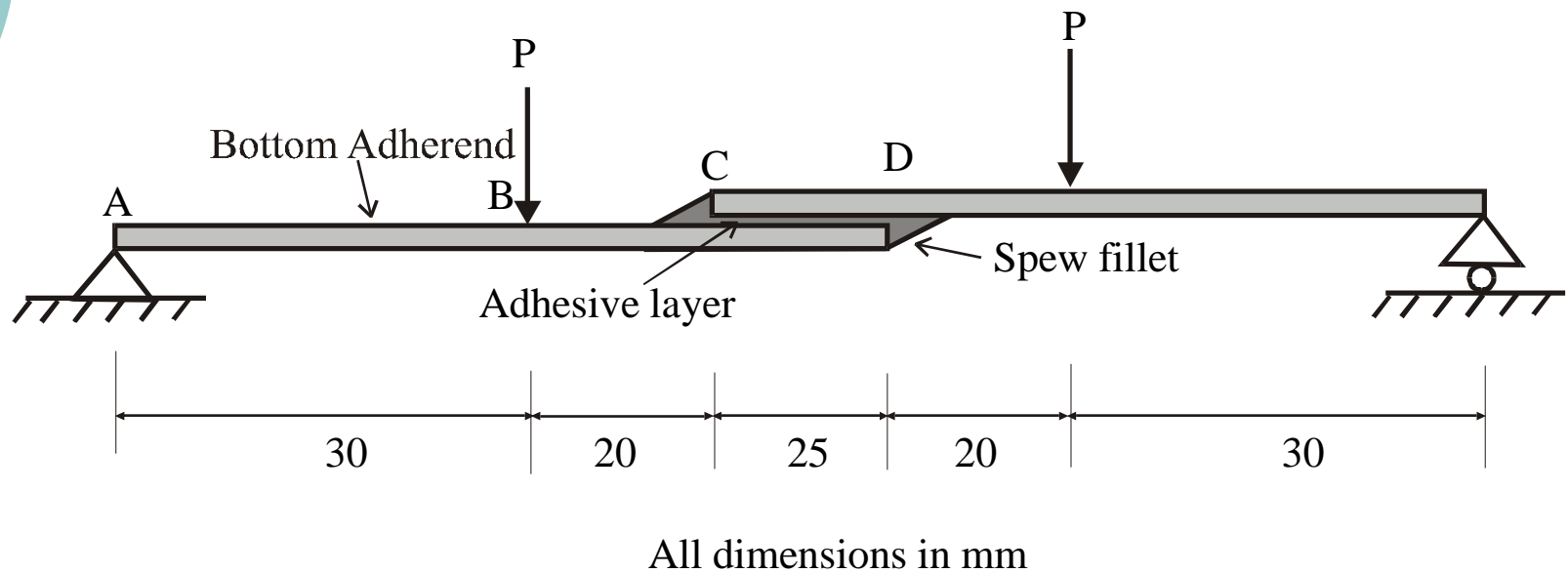


# A broken adherend from the tensile test

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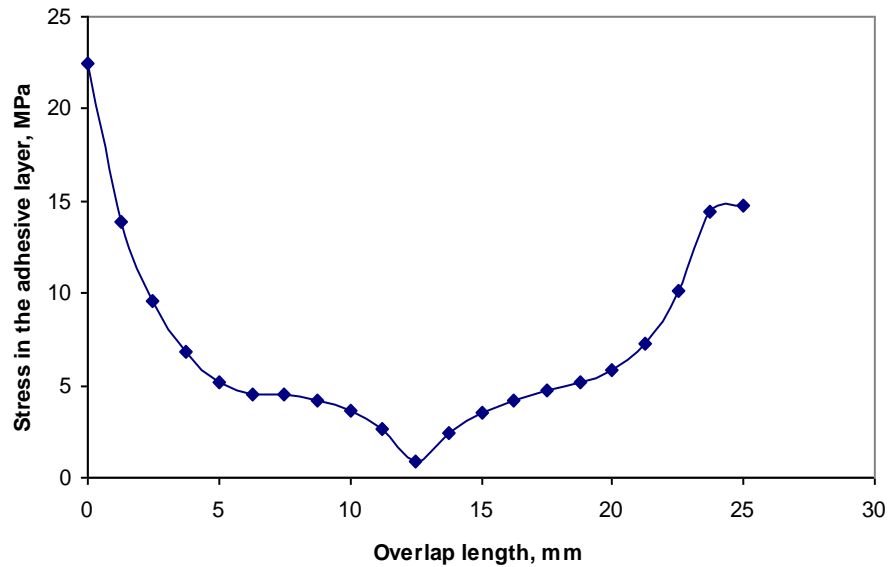


# Lap shear performance in quas-static loading under bending loading

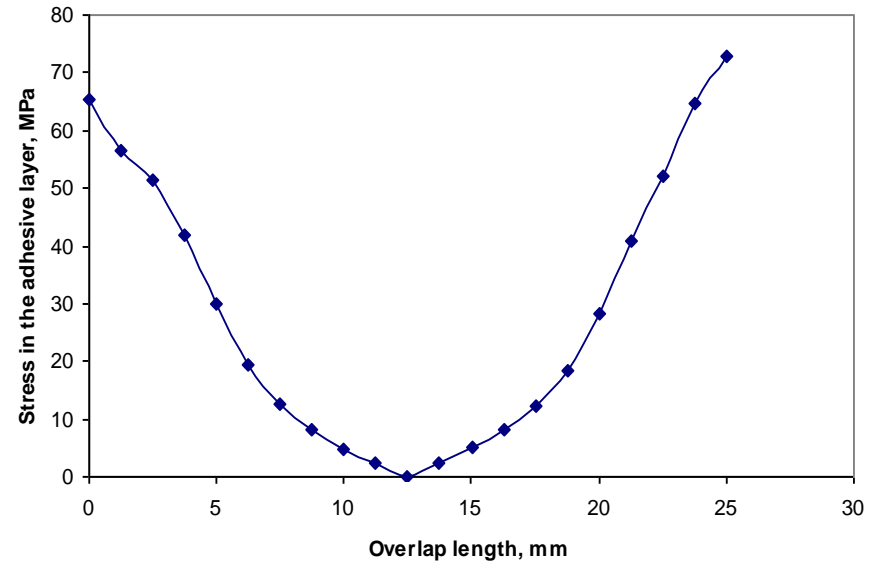


# Stress distributions in the adhesive layers under bending loading

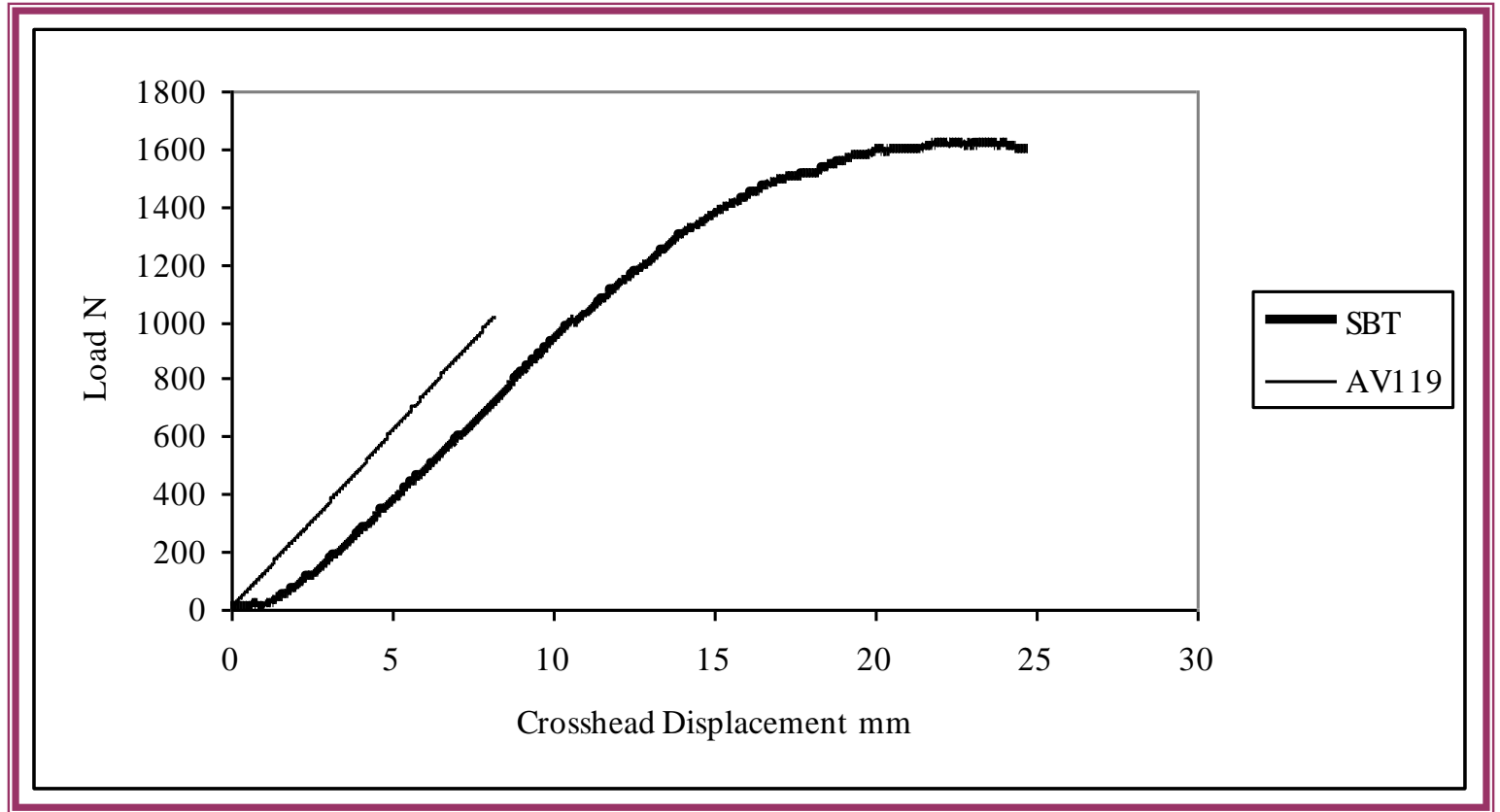
○ SBT



○ AV119



# Comparison of curves (results) from bending loading

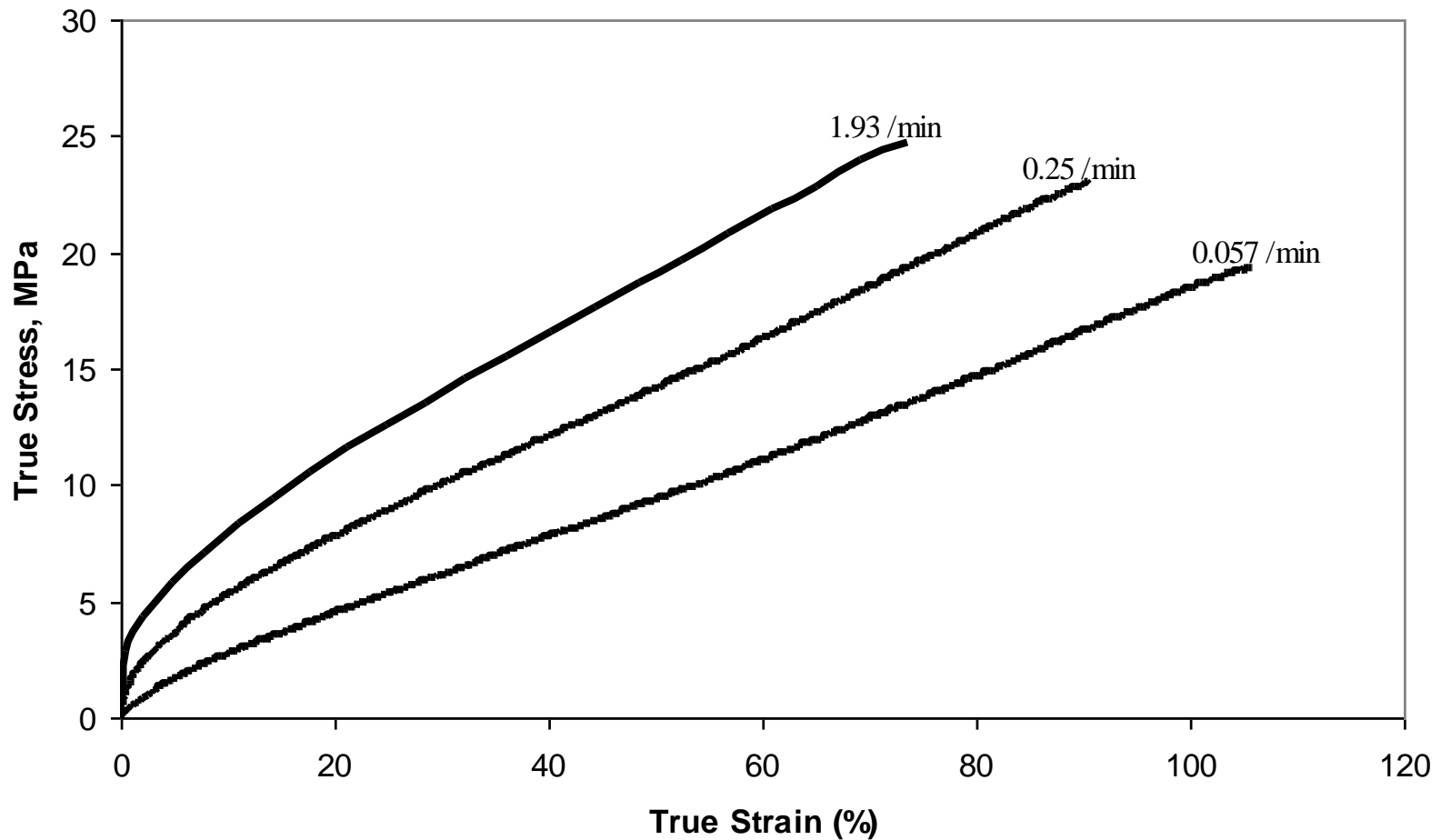


A broken adherend UTS= 2000 MPa

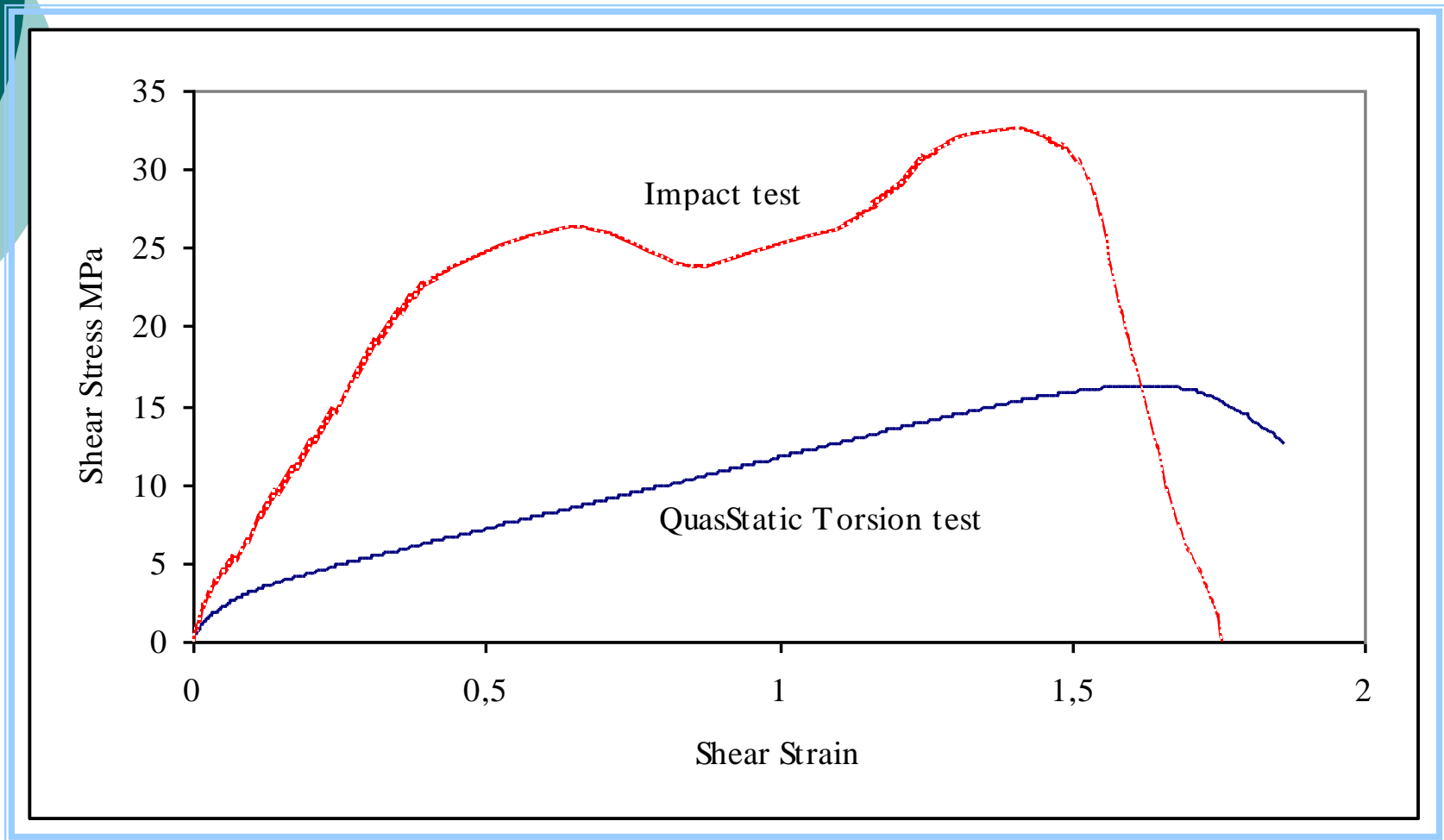
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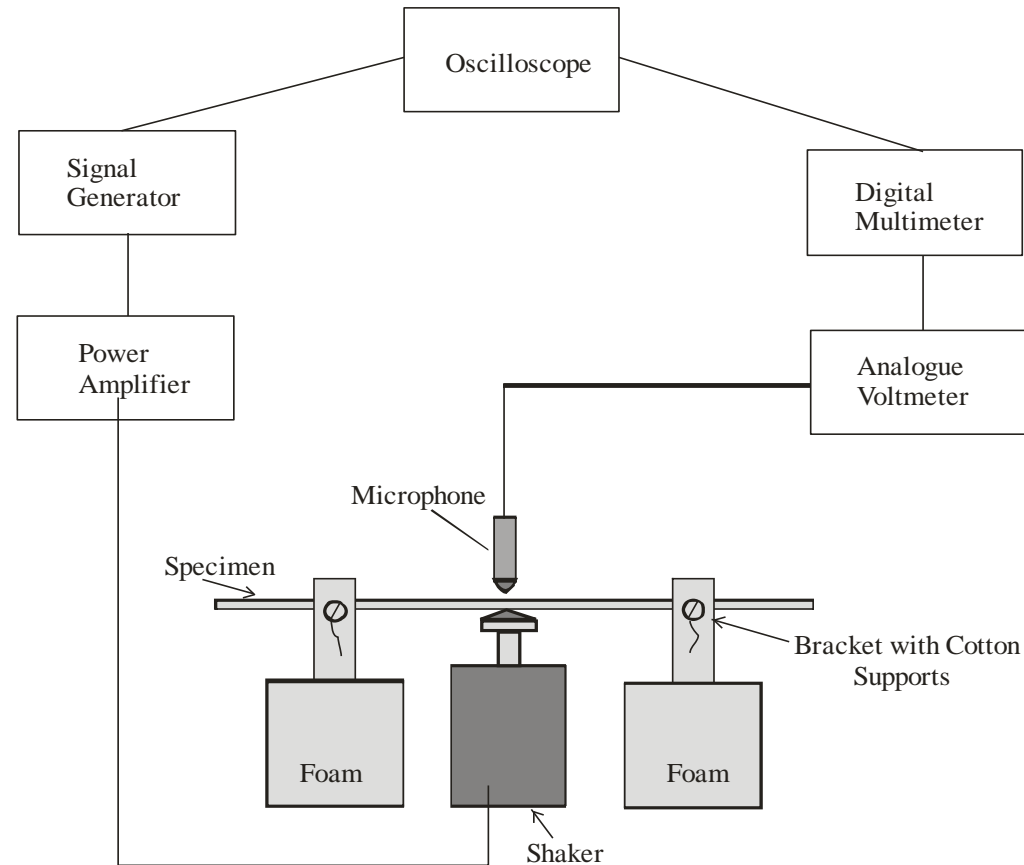
# SBT stress-strain variation curve with strain-rate variation at room temperature using a laser extensometer



# Comparison of quasi-static and impact test

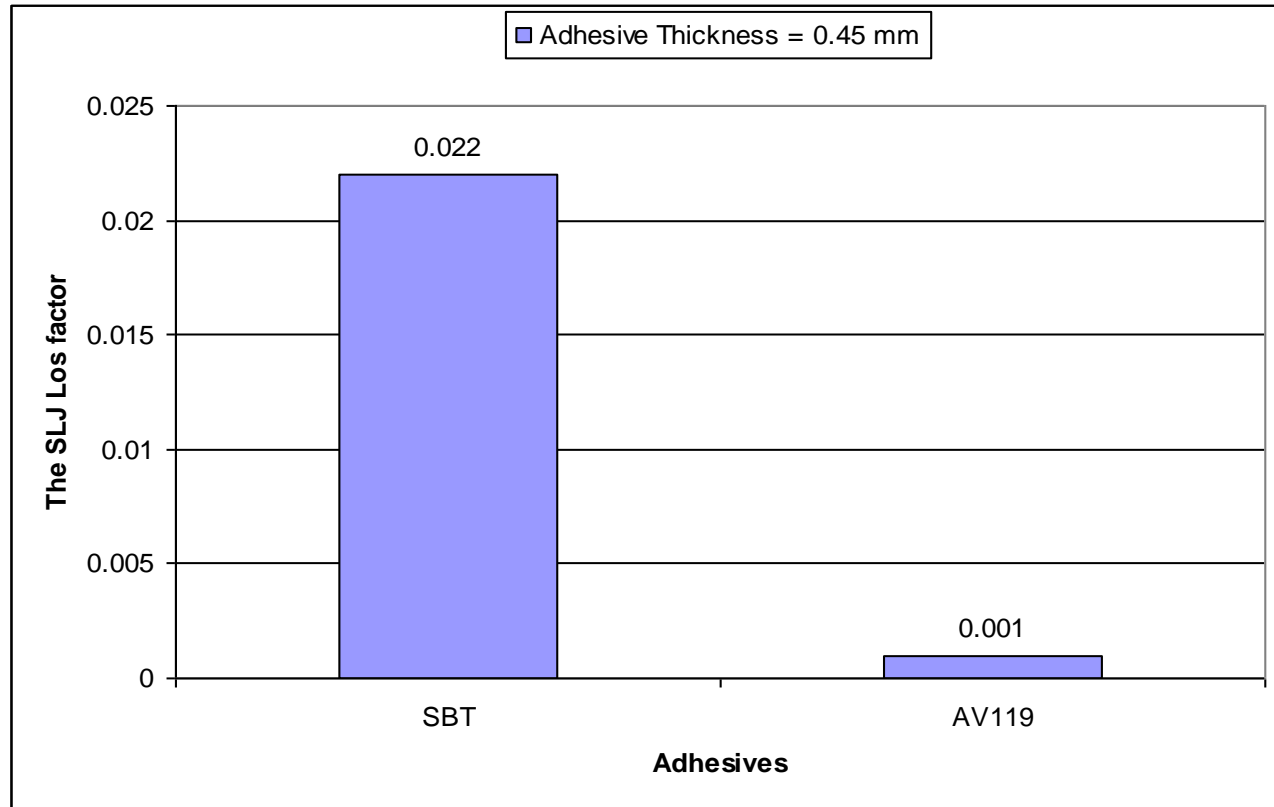


# Vibration set-up dynamic properties of materials





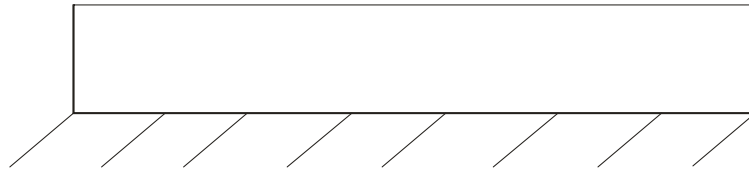
# Damping performance of the adhesives



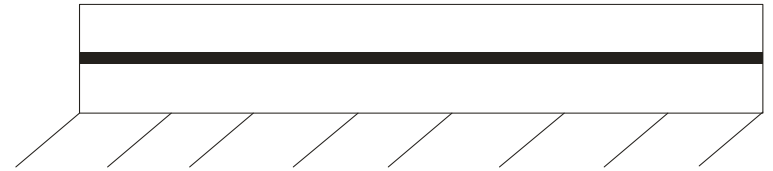
# Benefiting from damping properties of polymers (Adhesives)

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Solid



Sandwich



# Conclusions

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- Mechanical behavior of materials plays an important part in the performance of aerospace structures
- Correct design data are vital for reliable predictions
- Aerospace components require careful design approach
- Sometimes relatively less strong materials provide much more strong structural systems (SBT, AV119)



# Thank you !

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